



16th AM0 Workshop

December 3-6, 2012

Takayama, Gifu, Japan



The background of the slide features a photograph of a coastal landscape from an aerial perspective, showing a mix of green land and blue water. The top half of the slide has a solid black bar containing the workshop information.

U.S. High Altitude PV Measurement

for Primary Calibration Standard

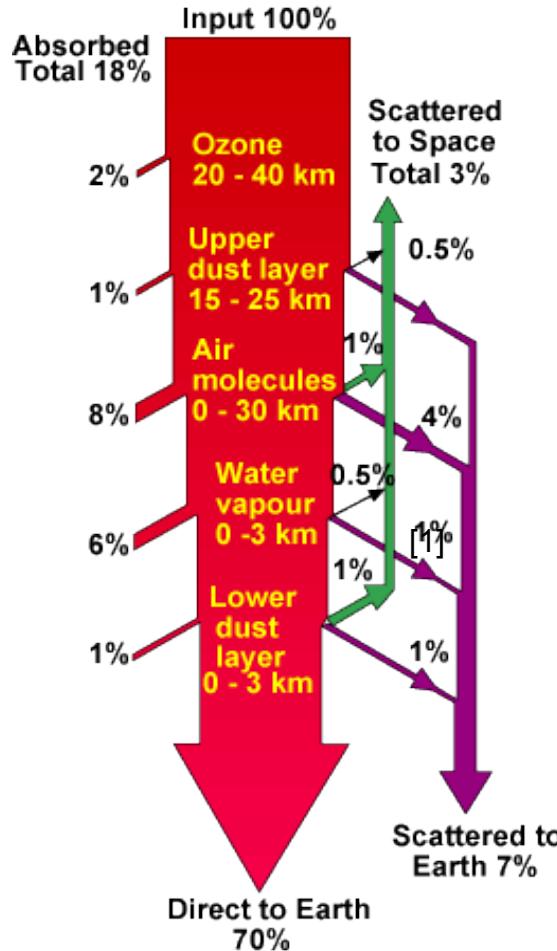
Certification

2008/10/11
18:37:57

Overview

- Background
- Current Capability
- Future Options
- Comparison of Learjet 25 and ER-2 Platforms

Background

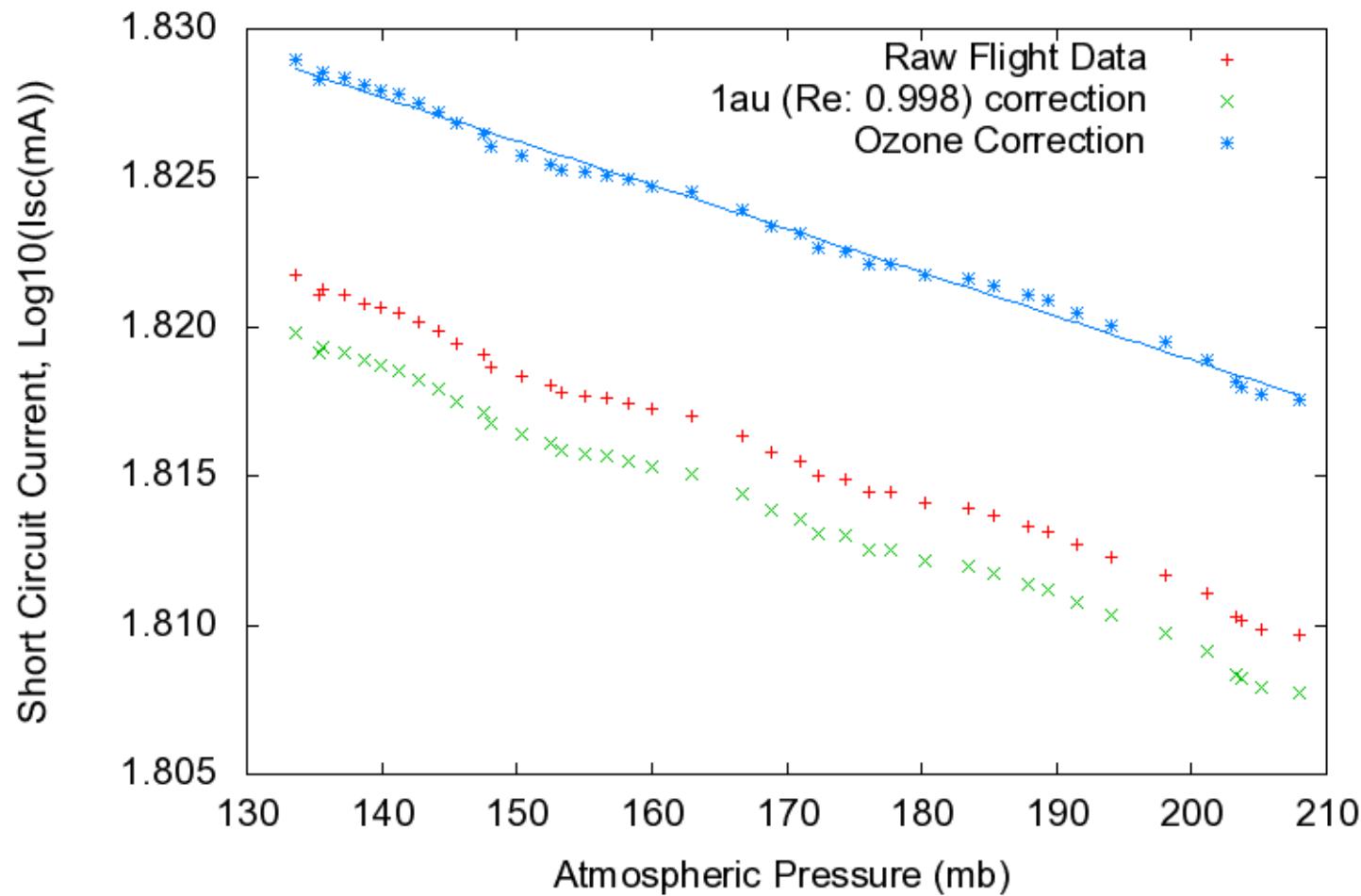


- AM0 primary calibration standards are required to calibrate laboratory solar simulators for measurement of space PV cells
- Ideally calibrations standards would be measured using a helium balloon platform above all but negligible absorbing and scattering components of the atmosphere (120,000 ft./36.6 km)
 - Maintaining funding for balloon measurement platforms have been problematic
- Currently U.S. standards are measured on board aircraft flying at the ceiling of their flight envelopes.
 - This method requires correction for ozone for 3J cells

Atmospheric Correction

- Isc data is corrected for current earth-sun distance and ozone levels.
 - Ozone correction based on satellite measurements of total ozone column density and a pressure dependent distribution model.
- Air Mass number calculated from pressure and sun declination angle.
 - Sun angle change is negligible during a 20 minute flight at local noon.
 - Pressure can be used directly.

Top Cell Data
Sun Altitude: 47.6 deg Total Ozone:331 d.u Trop: 210mb
Y-Int 1.8483 +/- 0.0003



Langley Plot Generated - Semi log plot of Isc vs. Air Mass is fit linearly and AM0 Isc extrapolated

Current Capability – Learjet 25



- Based at NASA – Glenn Research Center (GRC) Cleveland OH
- High Altitude testing has been conducted since 1963
- Operational Ceiling – 45,000 ft. (13.7 km) – 50,000 ft. (15.2 km) ASL
- Good correlation with balloon (1J, 3J) and shuttle tests (1J)
- Standard deviation in I_{sc} (AM0) ~ 0.75% for top cell, ~ 0.5% for middle and bottom cell
- I_{sc} measurement for all cells, I-V curve for limited number of cells

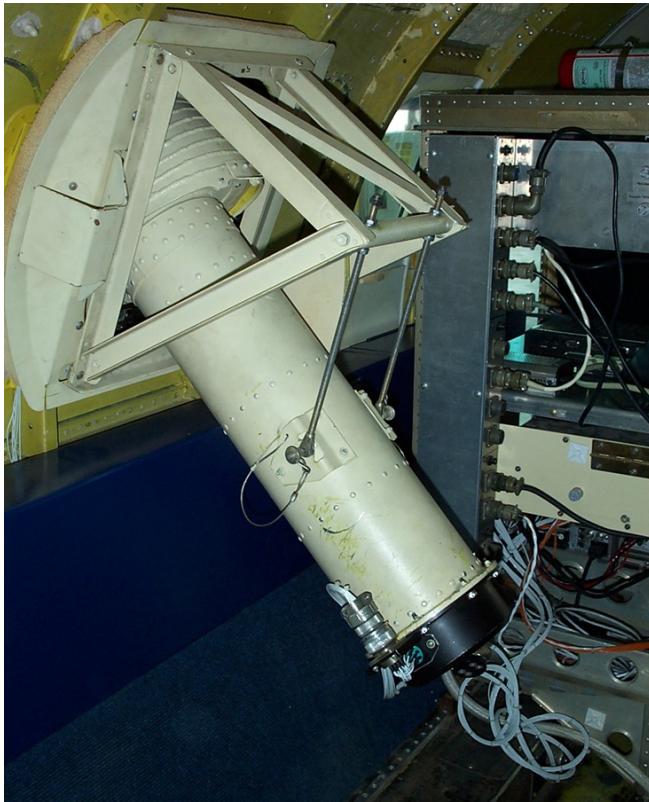
Method – Flight Plan

- Fly along 45th Parallel
- Ascend to 48,000 – 50,000 ft. (14.5-15 km) ~AM 0.2
- Data collection starts at local solar noon \pm 10 min
- Descend at 1000-1500 ft./min (300-450 m/min) to 30,000-35,000 ft. (9 -11km) while collecting data



Collimating Tube and Cell Fixture

- Temperature controlled within $\pm 1^{\circ}\text{C}$
- Angle can be set between 20-50° from horizontal



Sun Sight

- Enables aircraft pilot to align collimating tube with sun
- Sight is calibrated to tube prior to flight



Future U.S. Capability Options

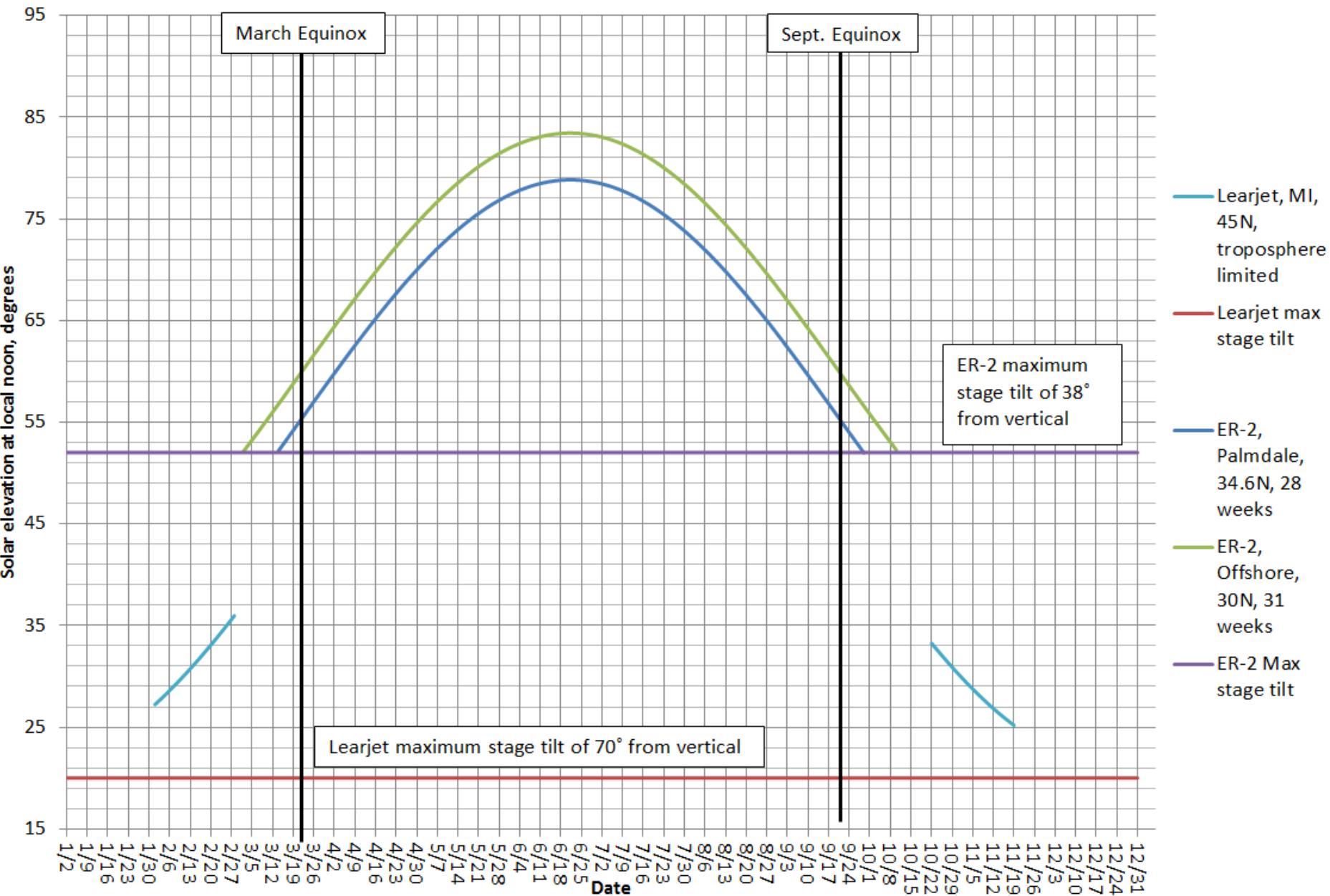
- NSCAP Balloon
 - NASA and DOD joint effort
 - Not currently funded
- NASA ER-2 High Altitude Aircraft – Dryden Flight Center
 - Under consideration

ER-2 Overview

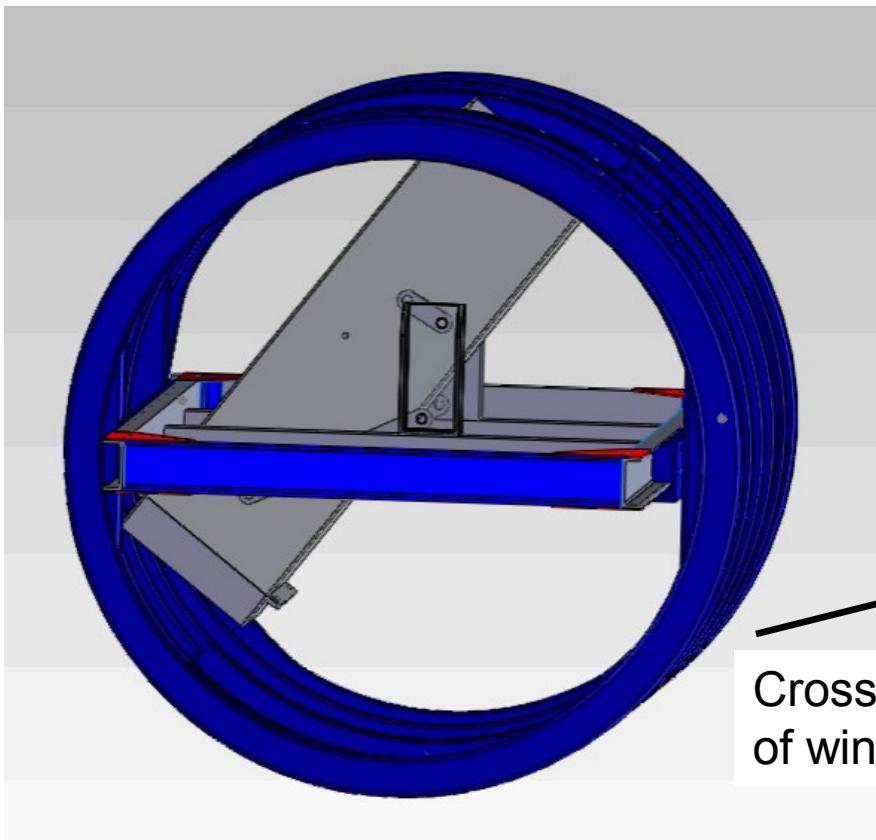
- The ER-2 AMO project would use NASA's ER-2 high altitude aircraft as a platform for solar cell measurements.
- The ER-2 can fly significantly higher (65,000 ft. vs. 48,000 ft. (19.8 km vs. 14.5 km)) than the Learjet-25.
- ER-2 flight parameters would compliment the Learjet 25 flight season and extend the overall testing season.
- Uses similar hardware and methods as the Learjet to accomplish the same goal – Obtain AMO performance data on solar cells.
- ER-2 would reduce the degree of ozone correction required and thus improve upon uncertainty for data collected at lower altitudes.
- Larger measurement stage area allows for increased size or number of samples.
- Use of 2 wing pods is a possibility.



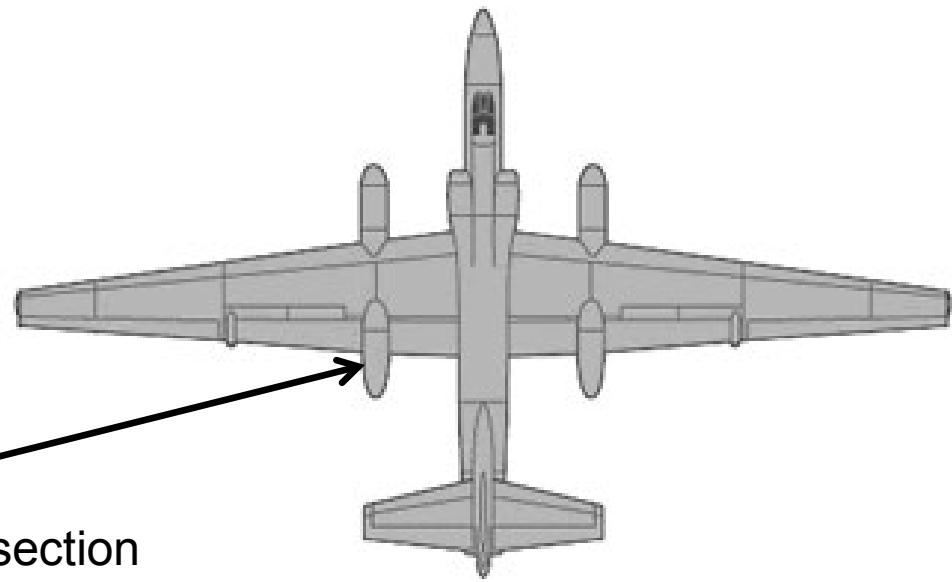
Normal flight opportunities for Learjet and ER-2 platforms



Concept of Collimation Tube Design



Cross section
of wing pod



38.2° maximum angle from vertical
3:1 length to diameter collimation ratio

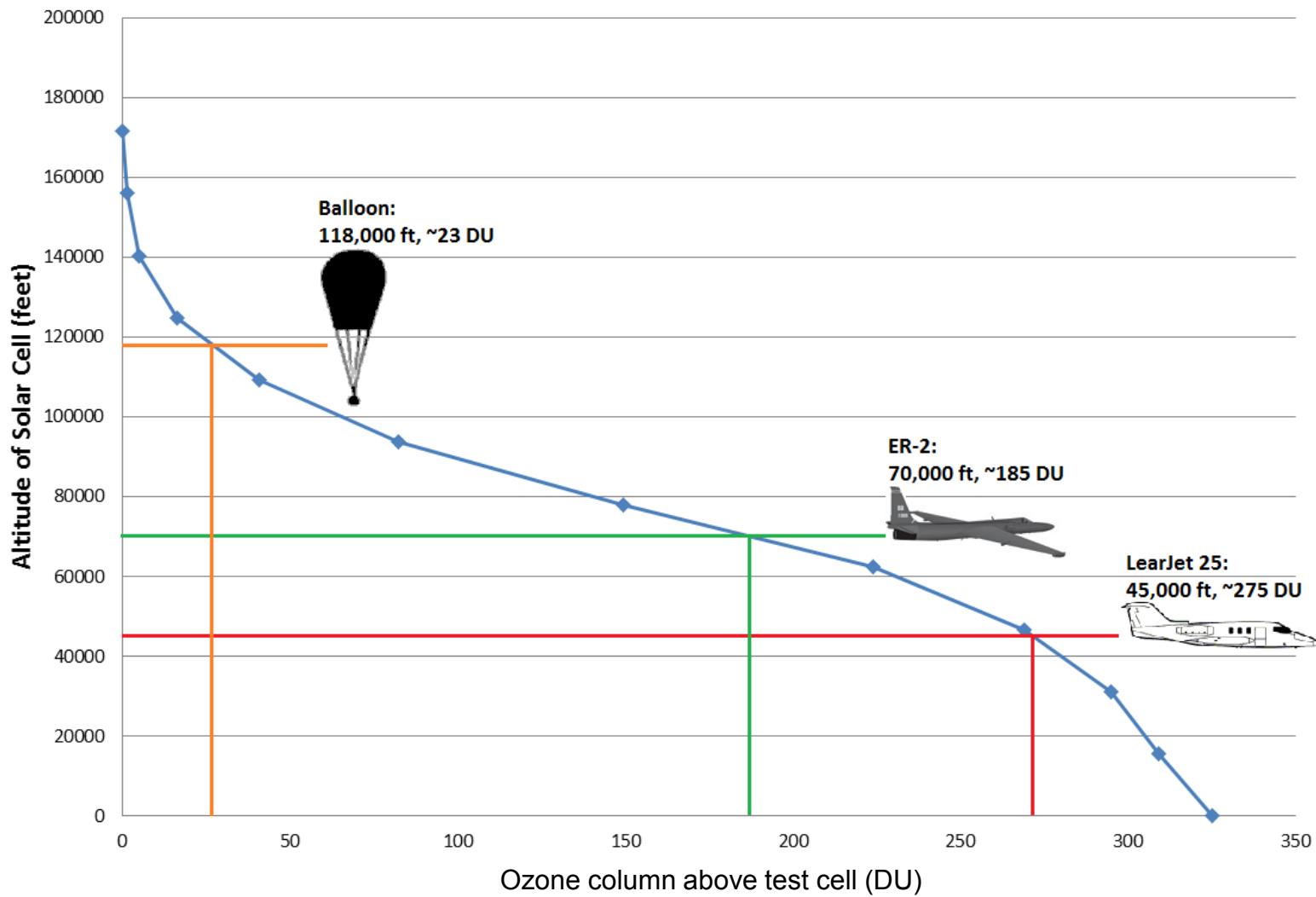
Comparison of ER-2 and Learjet Platforms

- Preliminary cost analysis indicates that costs per standard will be similar for either platform under certain conditions
- Factors that affect standard measurement costs per piece
 - Number of cells being measured
 - Number of flights required
 - Time of year

Cell capacity comparison

flight platform	Test area, cm	NSCAP 2 x 2 population	NSCAP 4 x 8 population	NSCAP 8 x 8 population	large area, 15.24 x 15.24 population
Learjet -25	11.4 \emptyset	6	2	1	0
ER-2 one pod	18.6 x 18.6	16	4	1	1
ER-2 two pods	2 x 18.6 x 18.6	32	8	2	2

~33% Reduction in Atmospheric Ozone vs. Learjet



Atmospheric Correction

Present Lear Analysis suggests Langley Plot extrapolation will be valid for ER2

- Less extrapolation (<½ air mass)
- Still need to correct for ozone
- Water vapor may be more of an issue for increased junctions
 - Need 4J tests/analysis
 - Need 6J tests/analysis (UV (ozone) and IR (water))
 - Smaller bandwidth increases sensitivity to absorption bands (O₂, H₂O, O₃)
 - Correction methods may be developed

Other NASA GRC PV Measurement Capabilities

- Triple source (xenon arc/tungsten filament/filtered tungsten filament) solar simulator
- Spectral response measurement
- Thermal cycling measurement
- Large area I-V curve measurement
- High concentration I-V curve measurement
- Low temperature I-V curve measurement
- Low intensity illumination measurement

